



The impact of inlet flow profile on the unsteady distortion characteristics of S-duct aero-engine intakes

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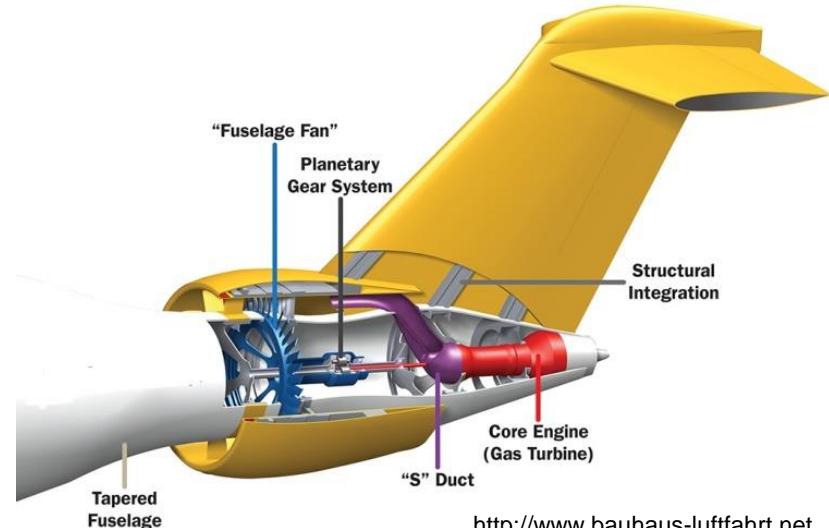
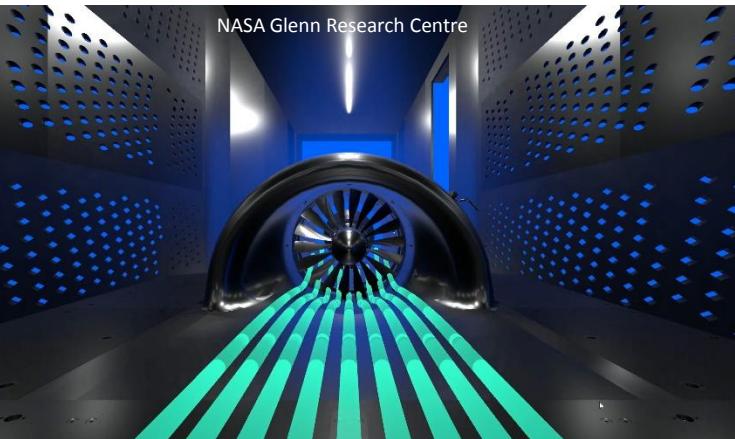


Agenda

- Introduction
- Research rationale
- Distortion screen design and verification
- Experimental capability
- Time-Resolved PIV at the AIP
- Results and discussion
- Conclusions

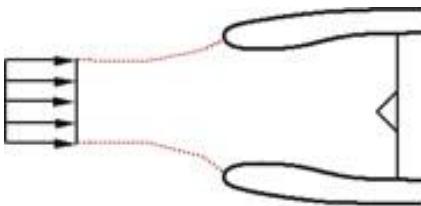


Research rationale

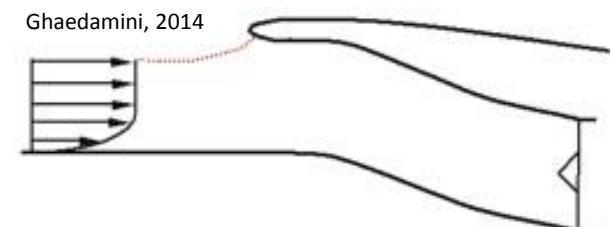


<http://www.bauhaus-luftfahrt.net>

- Synthetic method to generate non-uniform flow profiles.
- Quantification of flow distortion for range of inlet profiles.
- Boundary layer ingestion configurations.
- Across-the-range intake operation.

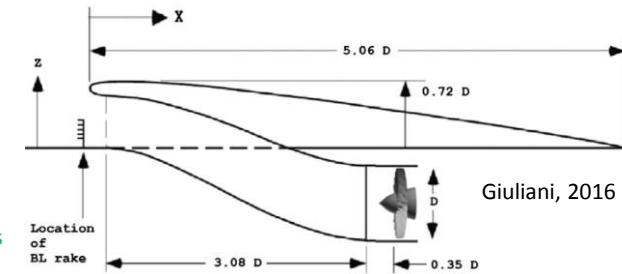
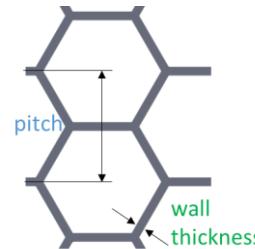
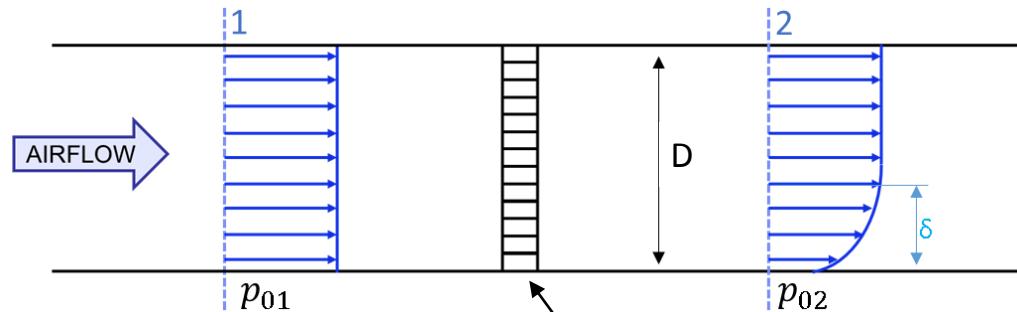


Conventional Inlet Flow



Boundary Layer Ingestion

Distortion screen design



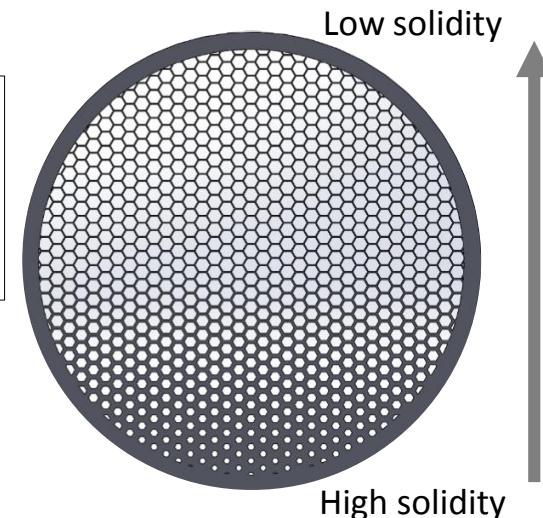
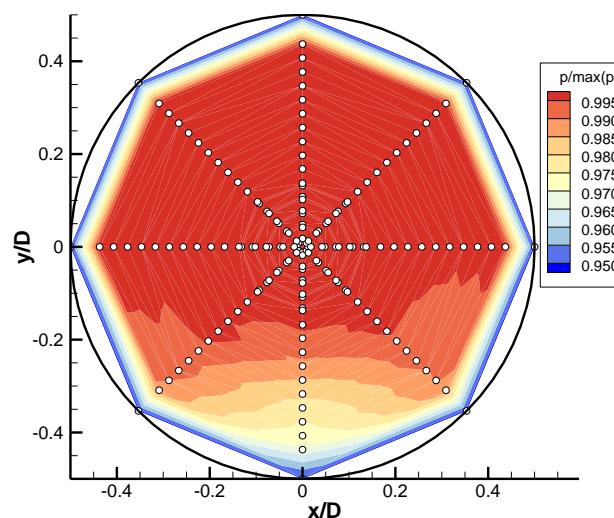
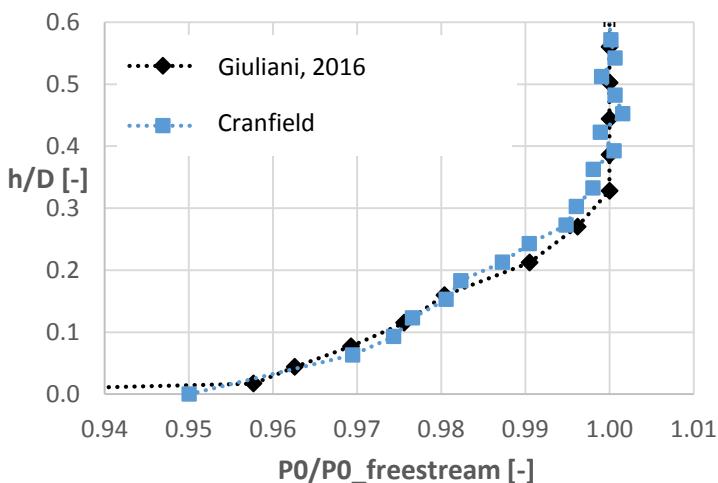
Target profile characteristics

δ/D	0.336
D (mm)	121.6
Inlet Mach	0.27

$$K = \frac{p_{01} - p_{02}}{\frac{1}{2} \rho V_2^2}$$

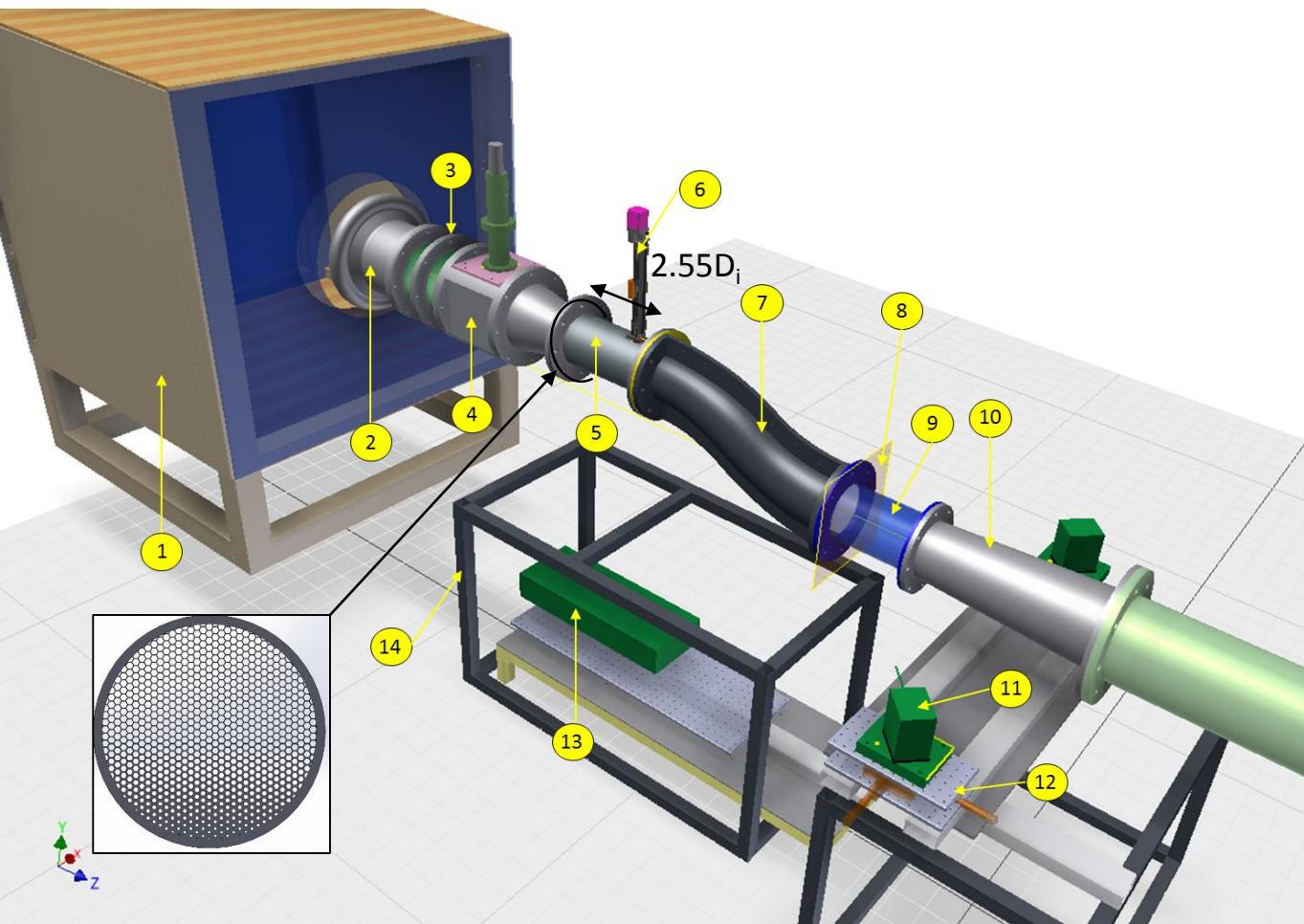
- Constant pitch.
- Variable wall thickness.

- Variable solidity final screen design.
- 3D printing production.





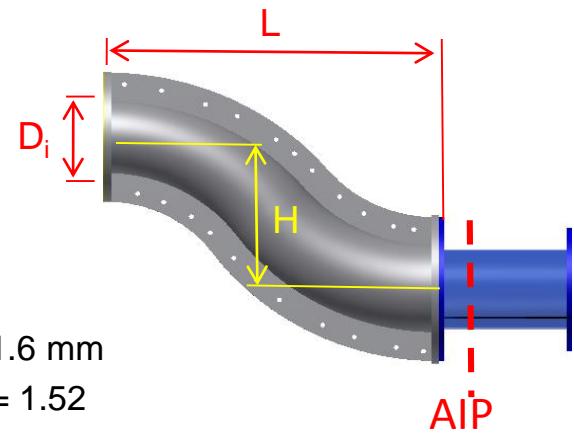
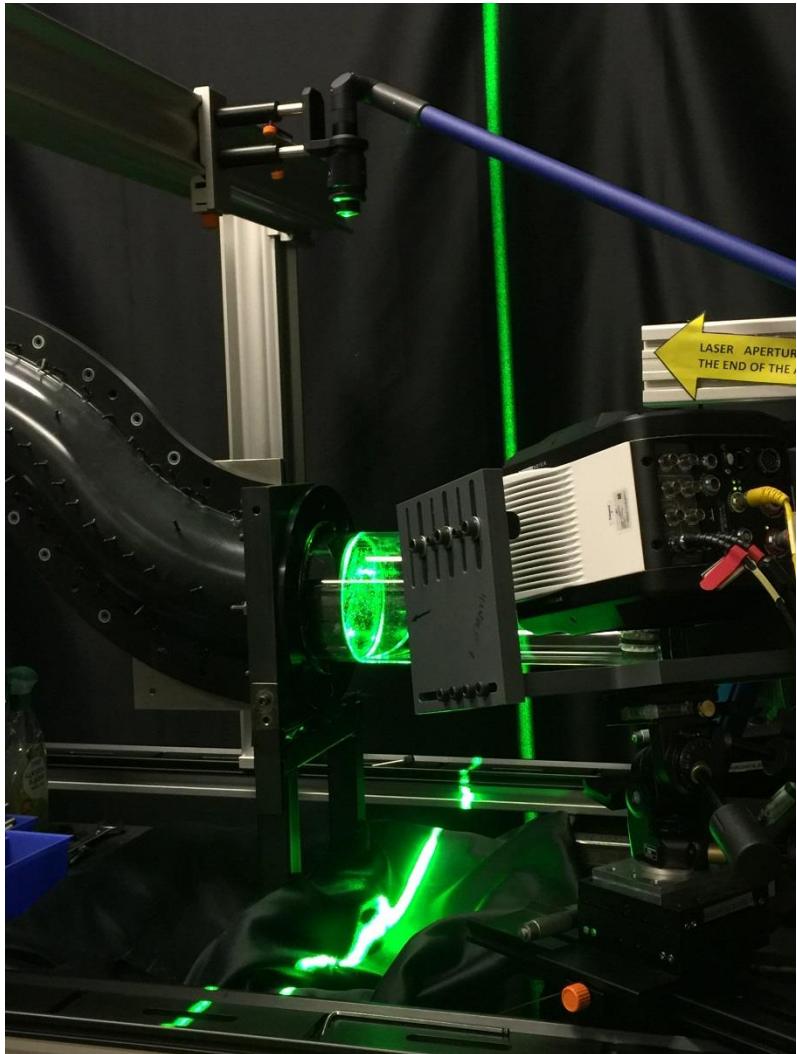
Experimental facility for bended intakes



- 1: Seeding chamber
- 2: Intake
- 3: Flow measurement section
- 4: Vortex generator section
- 5: Straight section
- 6: Inlet traverse station
- 7: S-duct
- 8: Measurement plane
- 9: Optical working section
- 10: Suction system
- 11: PIV camera
- 12: Camera traverse system
- 13: Laser
- 14: Support system

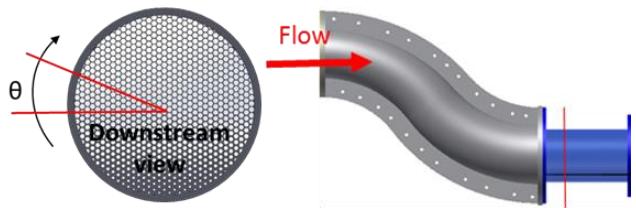


Time-Resolved PIV specifications



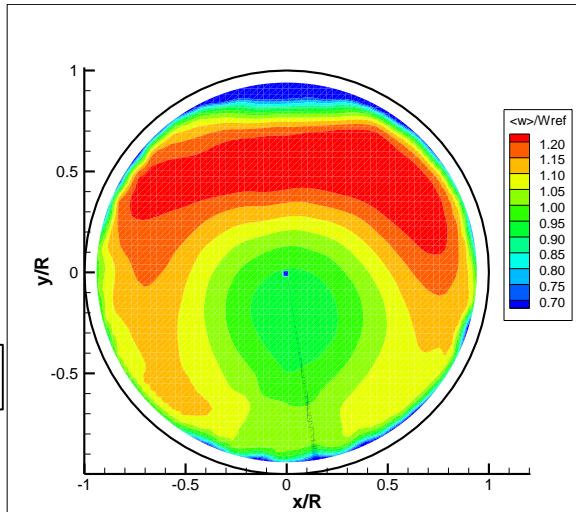
- **S-duct specifications**
 - Inlet diameter - $D_i = 121.6 \text{ mm}$
 - Aspect ratio - $A_{out} / A_{in} = 1.52$
 - Vertical offset - $H / D_i = 2.44$
 - Length - $L / D_i = 4.95$
 - Inlet Mach - $M_i = 0.27$
 - Inlet Reynolds - $Re_D = 7.48e+5$
 - Screen distance from inlet - $L_s = 2.55D_i$
- **PIV measurement plane**
 - AIP diameter - $D_{AIP} = 150 \text{ mm}$
 - Distance from S-duct outlet – $0.41D_{AIP}$
 - AIP Mach - $M_{AIP} \approx 0.21$
 - Velocity rep rate = 4 kHz
- **Time resolved stereo PIV (TR-S-PIV)**
 - Spatial resolution: $\sim 2.3 \times 2.3 \text{ mm}$
 - $\sim 3,000$ velocity vectors across $D_{AIP}=150 \text{ mm}$
 - Fully synchronous data across the plane
 - 3-component velocity vector at cross-flow plane
 - Software: LaVision Davis v8.3

Time-resolved PIV – out of plane velocity

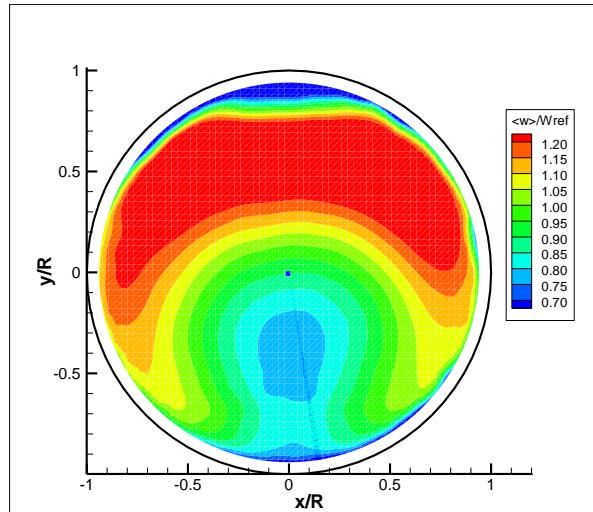


Time averaged out of plane velocity

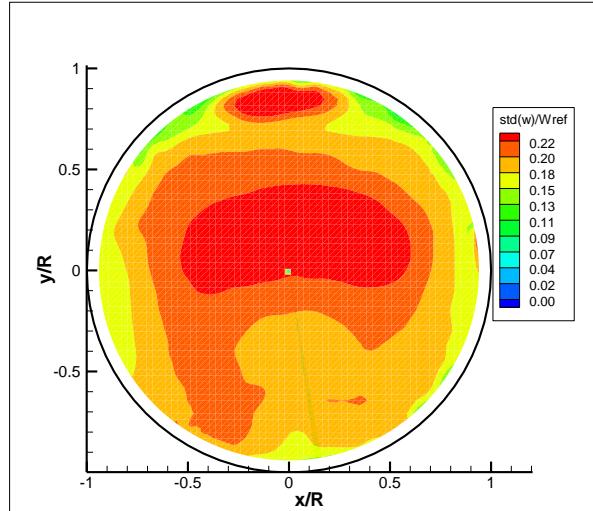
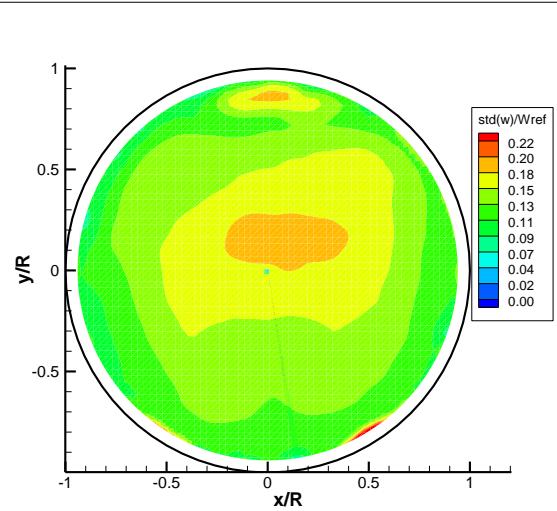
$\delta/D=0.04$ - Clean



$\delta/D=0.336, \theta = 0$ deg

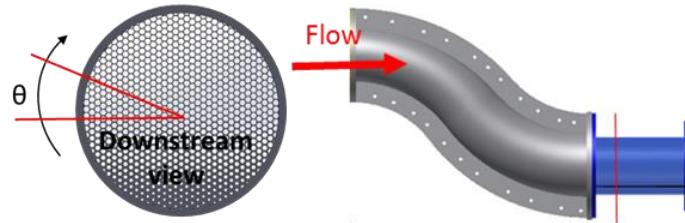


Out of plane velocity unsteadiness

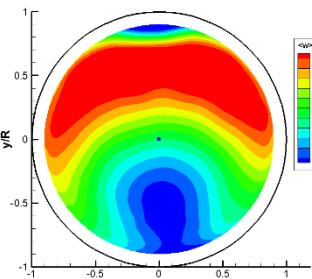


Time-resolved PIV

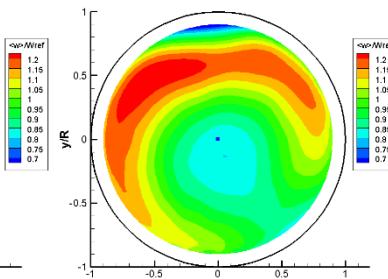
Time averaged velocity components



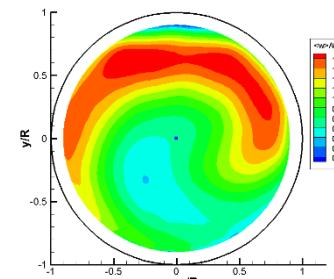
$\delta/D=0.336, \theta=0^\circ$



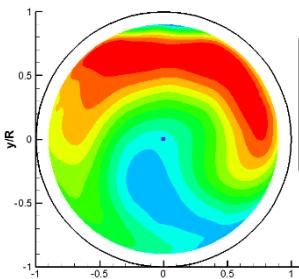
$\delta/D=0.336, \theta=45^\circ$



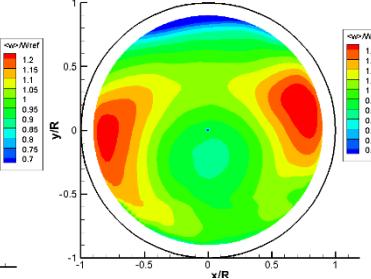
$\delta/D=0.336, \theta=90^\circ$



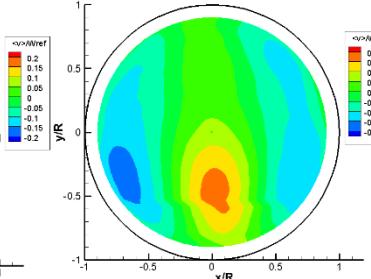
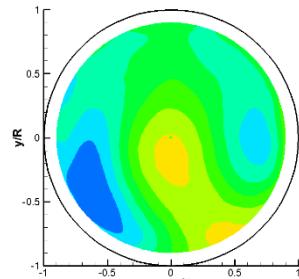
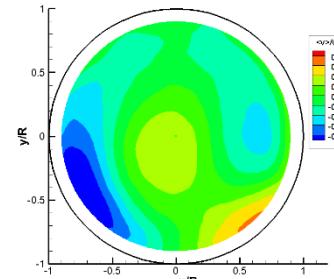
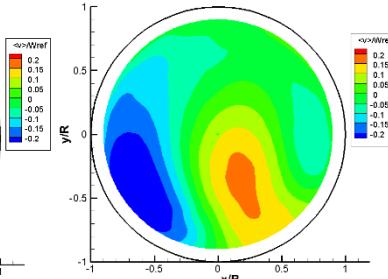
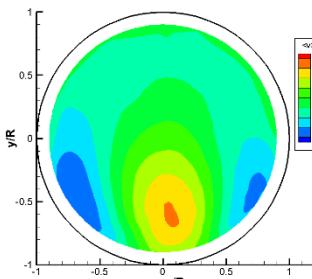
$\delta/D=0.336, \theta=135^\circ$



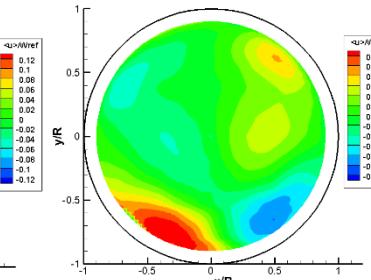
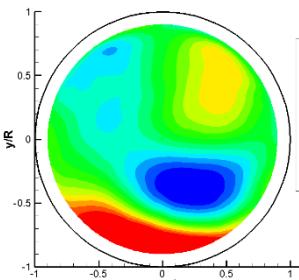
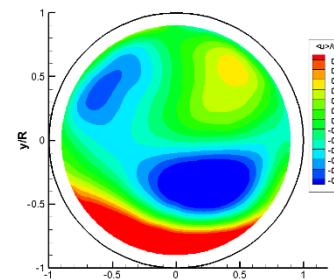
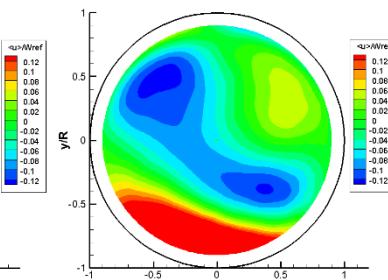
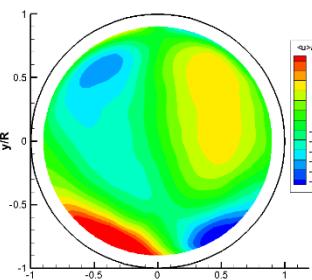
$\delta/D=0.336, \theta=180^\circ$



w/w_{ref}

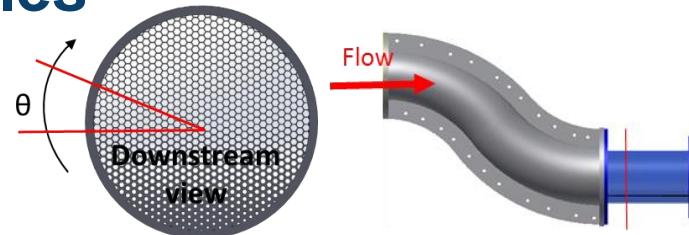


v/w_{ref}

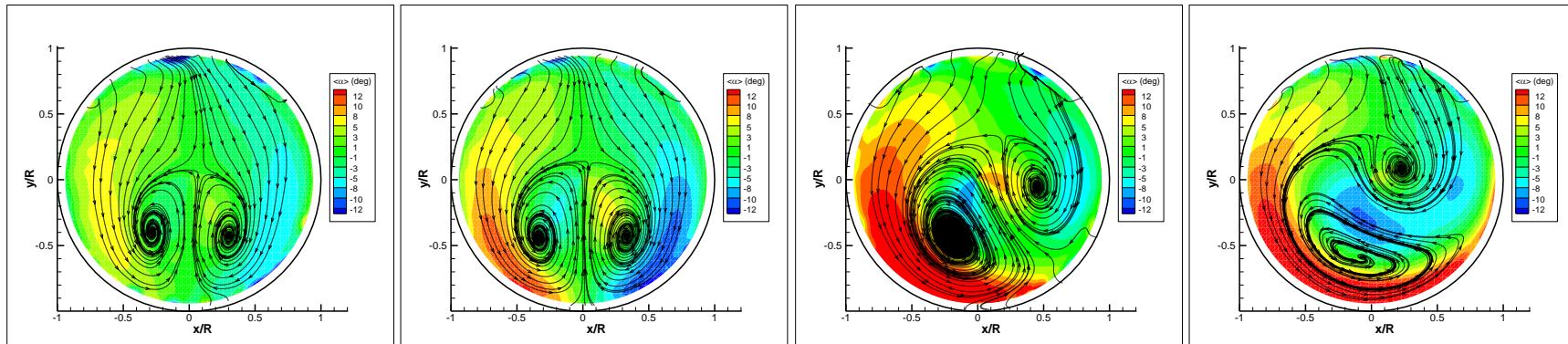


Time-resolved PIV – swirl angles

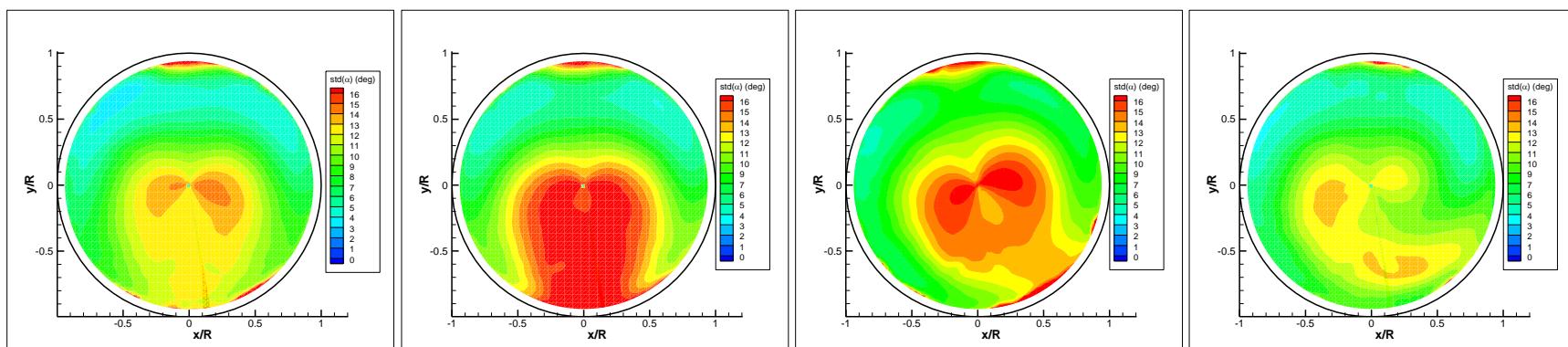
Swirl Angle



Time-averaged Swirl Angle



Standard Deviation



$\delta/D=0.04$ - Clean

$\delta/D=0.336, \theta = 0^\circ$

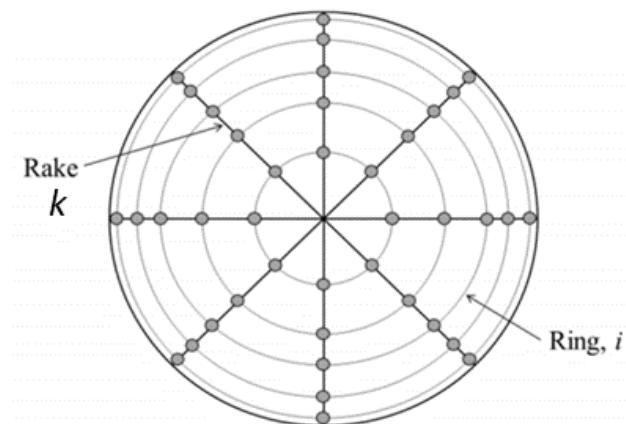
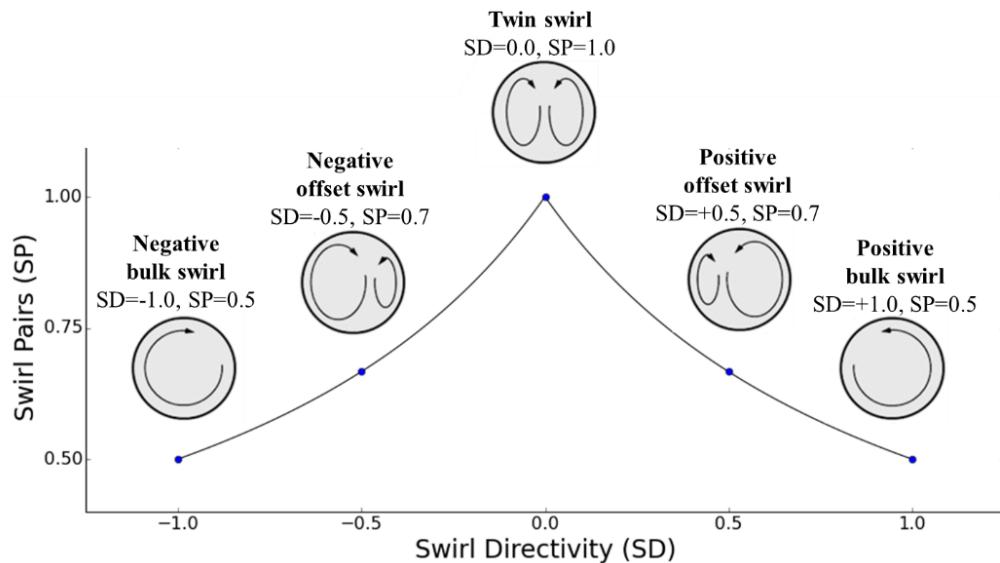
$\delta/D=0.336, \theta = 45^\circ$

$\delta/D=0.336, \theta = 90^\circ$

Swirl distortion

SAE swirl distortion descriptors

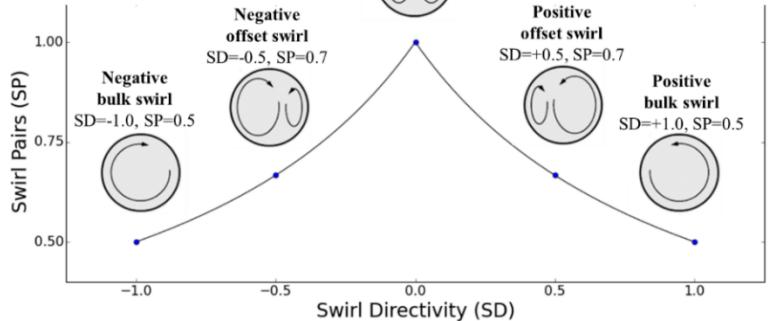
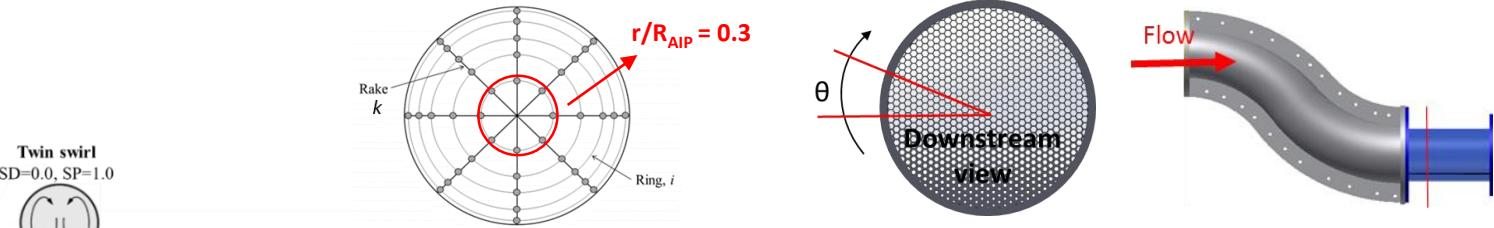
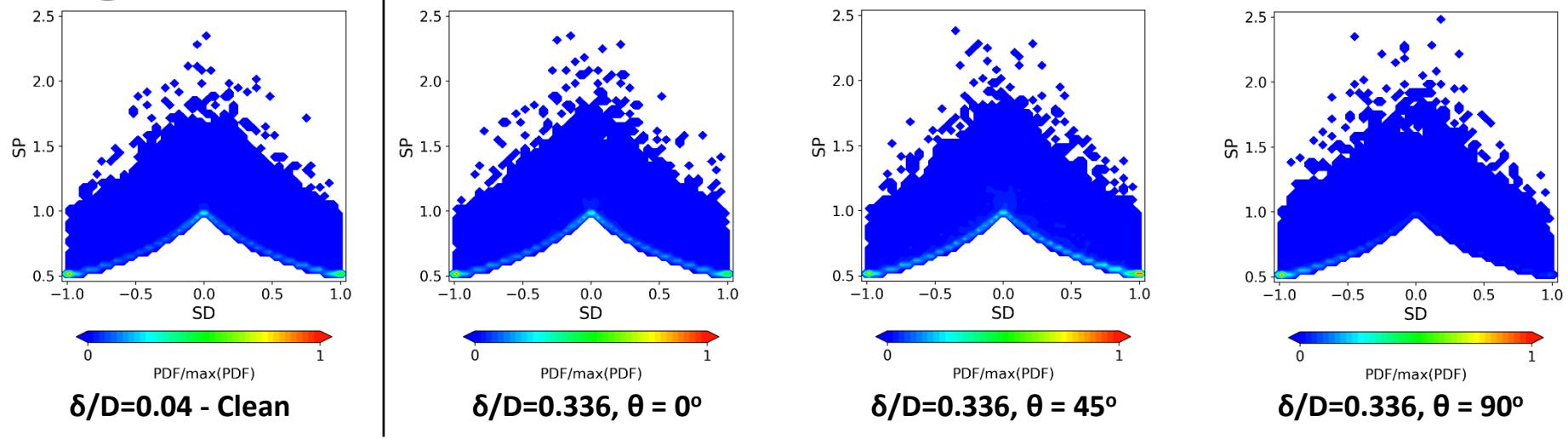
- Evaluated at rings and rakes
- Swirl Intensity (SI) quantifies the swirl levels
- Swirl Pairs (SP) and Swirl Directivity (SD) characterize the swirl pattern



$$SS_{i,k}^+ = \frac{1}{\theta_{i,k}^+} \int_{\theta_{i,k}^+} \alpha(\theta)_i d\theta \quad SS_{i,k}^- = \frac{1}{\theta_{i,k}^-} \int_{\theta_{i,k}^-} \alpha(\theta)_i d\theta$$

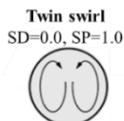
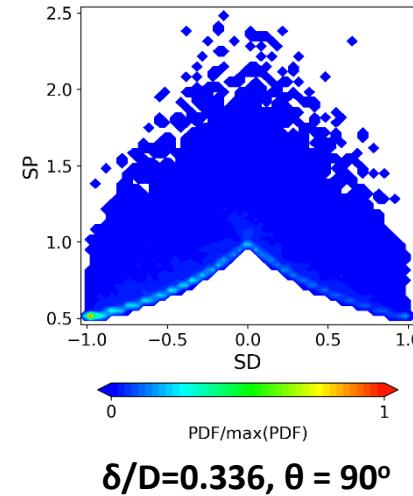
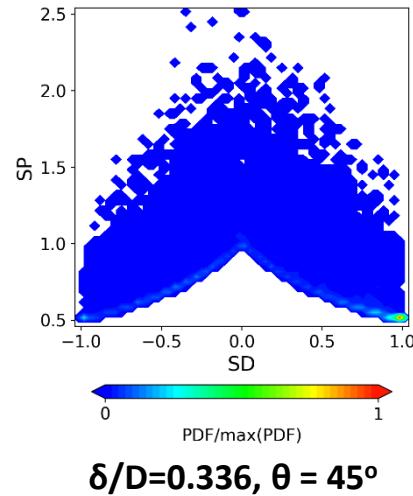
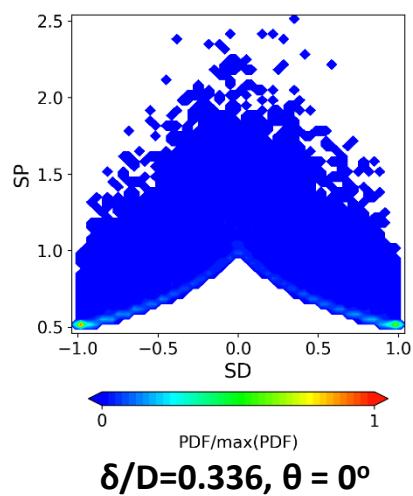
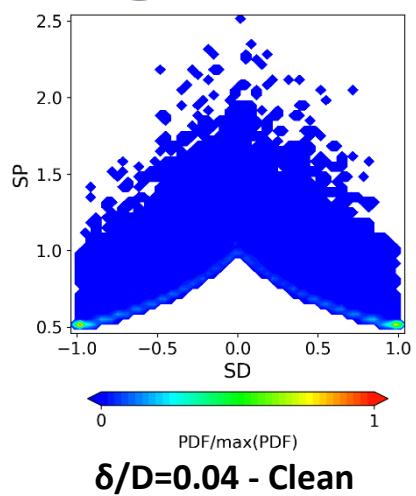
(SAE, 2007)

Time-resolved PIV – Swirl descriptors – Hub



- No pattern variation between clean, 0 and 45 deg
- Negative bulk swirl locking for 90 deg case

Time-resolved PIV – Swirl descriptors – Mid-span

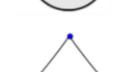


Twin swirl
 $SD=0.0, SP=1.0$

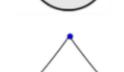
Negative bulk swirl
 $SD=-1.0, SP=0.5$



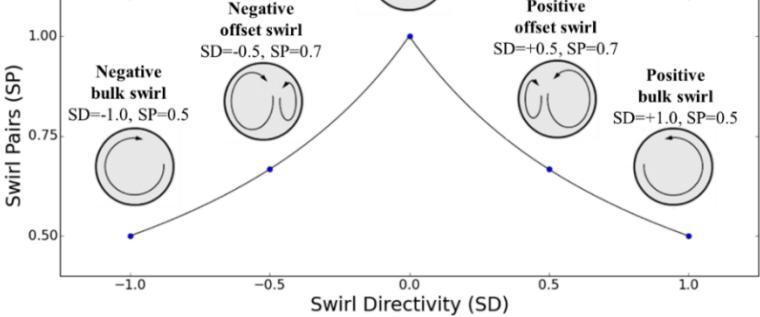
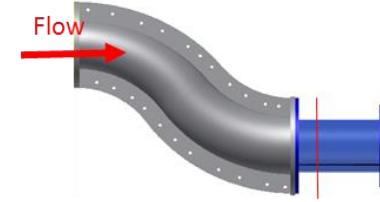
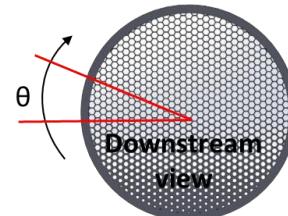
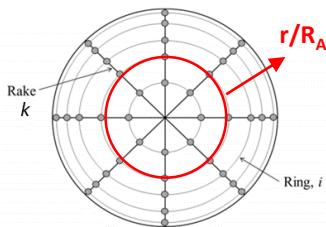
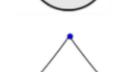
Negative offset swirl
 $SD=-0.5, SP=0.7$



Positive offset swirl
 $SD=+0.5, SP=0.7$

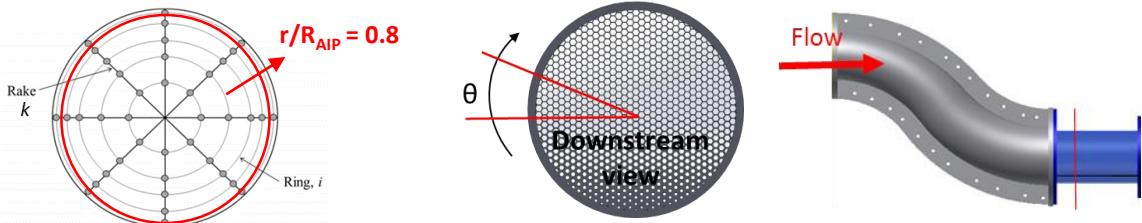
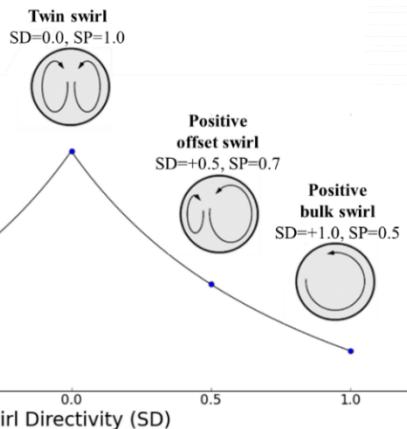
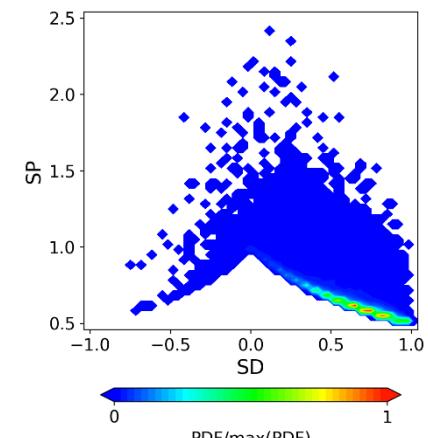
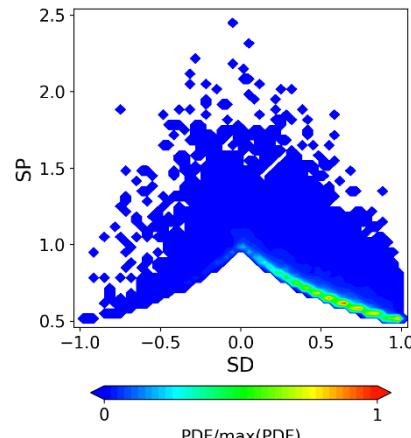
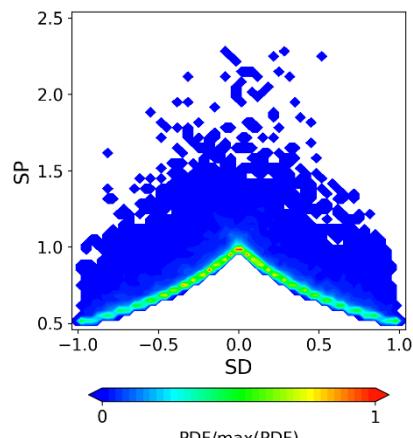
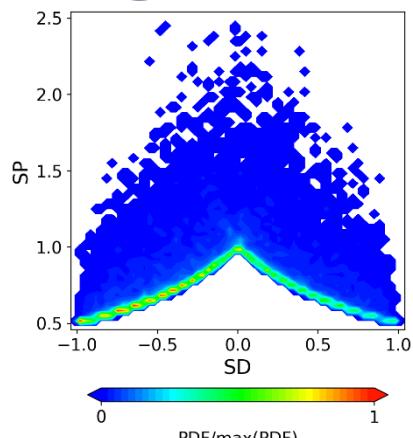


Positive bulk swirl
 $SD=+1.0, SP=0.5$



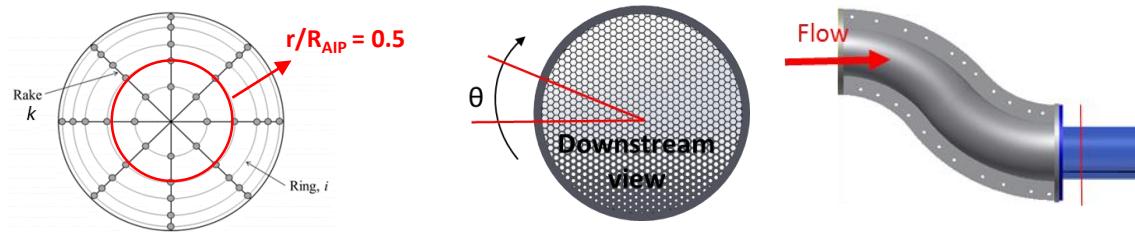
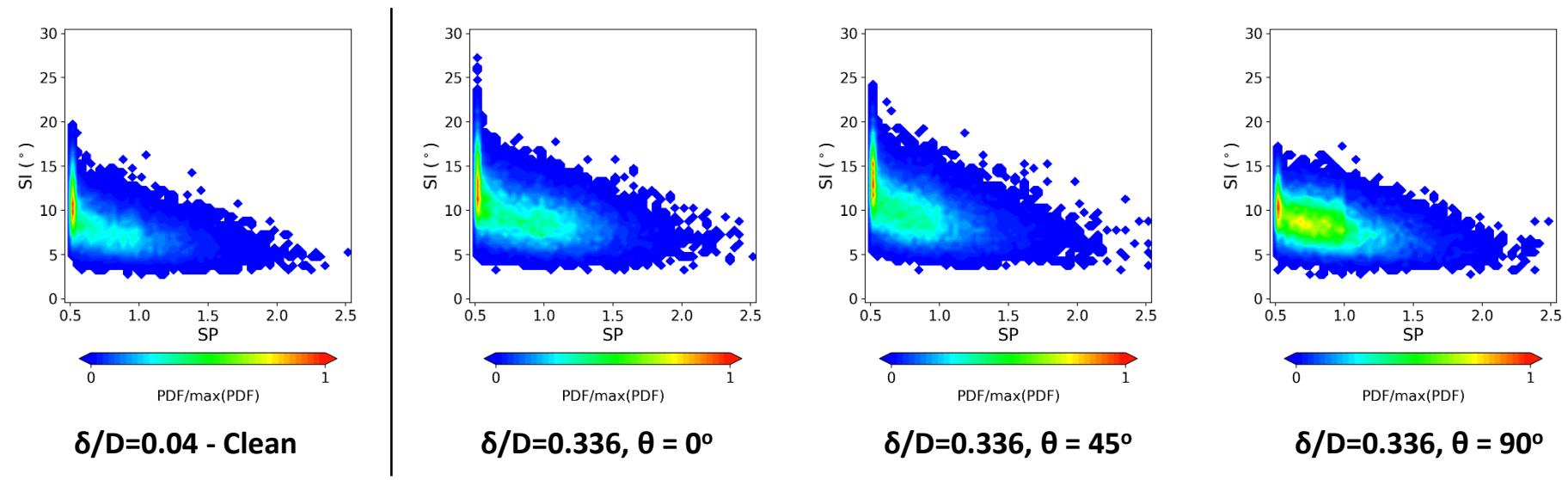
- No pattern variation between clean and 0 deg case
- Positive bulk swirl locking for 45 deg case
- Predominant negative bulk swirl for 90 deg case

Time-resolved PIV – swirl descriptors – Tip



- Promotion of twin swirl events at 0 deg
- Shift to positive swirl developing at 45 deg
- Predominance of positive swirl at 90 deg

Time-resolved PIV – Swirl descriptors – Intensity



- $\delta/D=0.336, \theta = 0^{\circ}$ - swirl intensity peak values increased by 40%.
- $\delta/D=0.336, \theta = 90^{\circ}$ - narrow range – intensification of bulk swirl events.



Conclusions

- Inlet pressure profile generation via flow distortion screens.
- Inlet flow profile characteristics enable off-design intake operating points.
- Inlet δ/D has notable impact on AIP velocity unsteadiness.
- Increase of swirl angle range – identification of non-uniform blade loading areas.
- Distortion screen has different effect on tip-hub swirl angle topology.
- Increase of swirl intensity up to 40% in negative bulk swirl topology.



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